

LUMOS

ESA

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→ THE EUROPEAN SPACE AGENCY

Cis-lunar Domain : Raising awareness and real cases



Memorandum on US Space Policy Directive 7

Consistent with the National Space Traffic Management Policy, PNT services will also play an important role in space traffic management and future applications in the Cislunar Service Volume, ...

2021



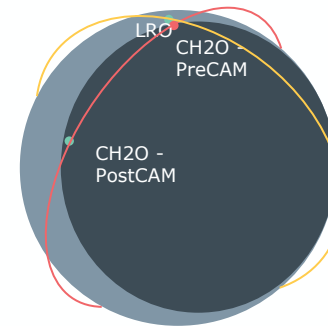
US National Cislunar Science & Technology Strategy

... demonstrate cost-effective space-based and Lunar surface sensors

2022



Chandrayaan-2 Orbiter (CH2O) performs an evasive manoeuvre to mitigate a close approach with NASA LRO



2021

the guardian

Space junk on collision course with the moon likely a Chinese rocket

“Mystery” object was not a SpaceX Falcon rocket upper stage ... maybe the third stage of a Chinese rocket

2022

Challenges

Difficult observation conditions from ground-based and low-Earth observers:

- Moon occultations
- Straylight and atmospheric scattering
- Weather dependency
- Limited diversity in observation geometry



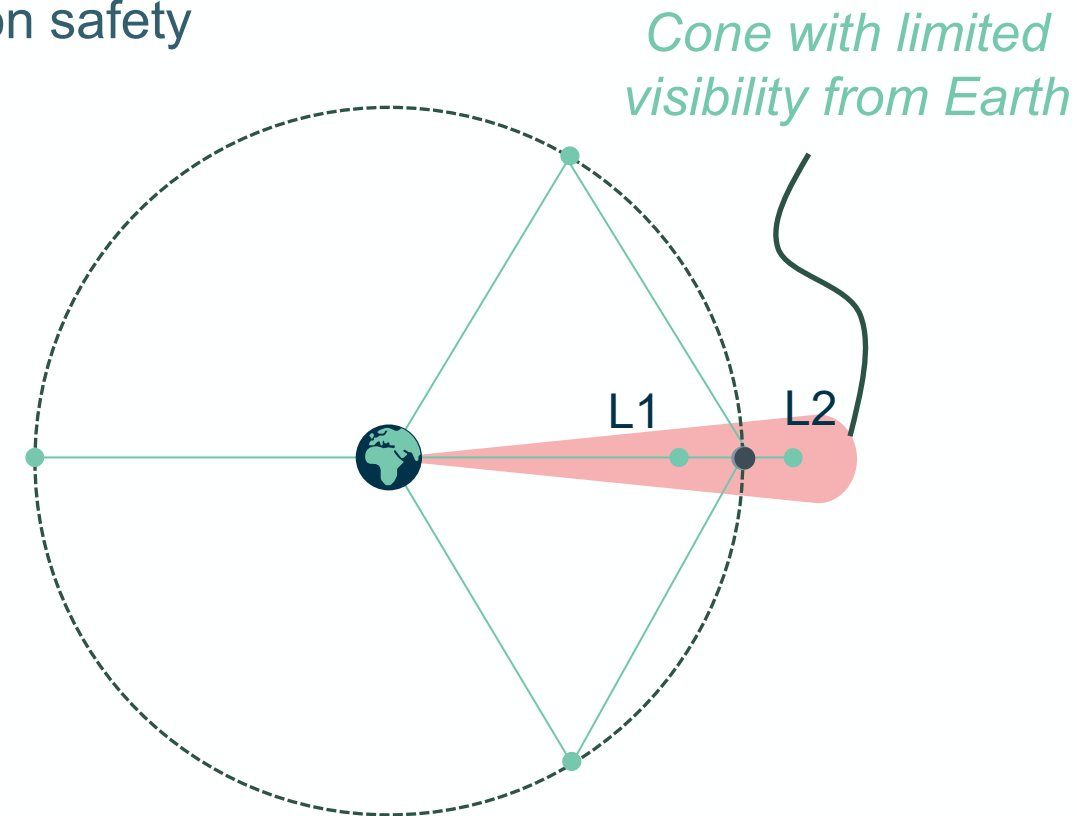
Improved geometries from optical payload in cis-lunar space

- Lunar missions require high Δv for launch and operations
- Trajectory should allow finding objects (high duty cycle and revisit period)

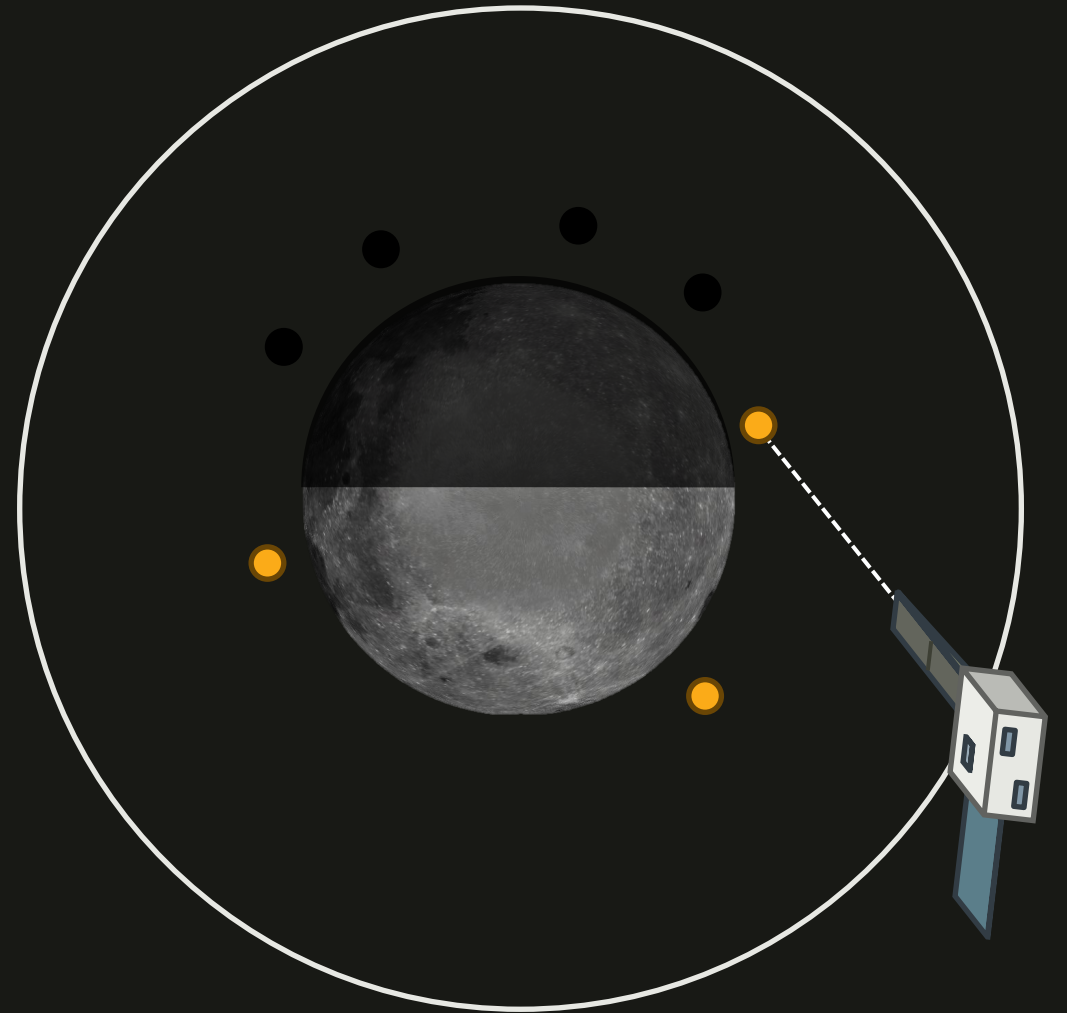
Mission Objectives

Objective: Demonstrate capability to detect, find, and track objects in cislunar orbits (Earth-Moon transfer, lunar, Lagrange point orbits)

- Objects: at least 1m (target 10 cm), in stable orbits or crossing operational orbits
- Enable collision avoidance operations and support mission safety
- Assess Earth/Moon/Heliocentric capture during disposal
- Predict Earth/Moon impacts
- Characterise objects (brightness)
- Backup navigation support for lunar missions e.g. in contingency situations

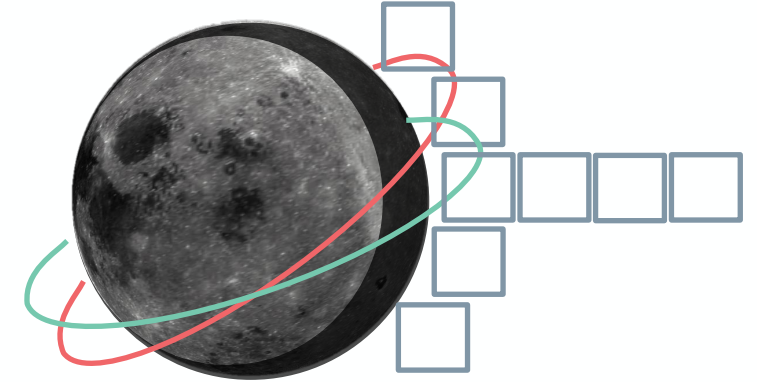


- Passive-optical payload in vis. range (based on heritage)
- Deployable “Moon-shield” for direct straylight reduction
- DRO 40.000 km (approx. circular 7 days period)
- Duty cycle: ~40% (SNR>threshold for higher lunar targets targets)
- Observation windows for low-lunar orbits (exclusion from terminator line)



Observation strategy

- Survey area of interest (all stable orbits - may be reduced further) in ~8 days
- Tracking and survey mode
- On-board processing for object detection (data reduction rate based on heritage software using differencing algorithm)



Support from ground-based telescope network:

- High-lunar altitude orbits may be observed from ground-based sensors

Launch

1. Launch into standard GTO
2. 7 days loitering (16 revolutions)
3. Apogee Raising Manoeuvre (ARM) to WSB
4. Up to three Deep Space Manoeuvres (DSM)
5. Lunar Gravity Assist (LGA)
6. DRO Insertion Manoeuvre (DIM)

Alternative: Direct insertion into WSB

